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PRINT DATE: 06/14/90

FAILURE MODES EFFECTS ANALYSIS (FMEA) — CRITICAL HARDWARE

NUMBER: 06-38-0403-X

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SUBSYSTEM NAME: ATCS - AMMONIA BOILER SYSTEM

REVISION: 2 06/13/90

PART NAME VENDOR NAME PART NUMBER VENDOR NUMBER

LRU :

AMMONIA BOILER SUB-SYSTEM

MC250-0005-0007

74716050

LRU : TEMPERATURE SENSOR, ABS

74715048-5 74716050

PART DATA

EXTENDED DESCRIPTION OF PART UNDER ANALYSIS: TEMPERATURE SENSOR, AMMONIA BOILER FAULT LOGIC.

QUANTITY OF LIKE ITEMS: 2 ONE/SYS. A OR B CONTROLLER; TWO PER VEHICLE

FUNCTION:

PROVIDES UNDER TEMPERATURE SIGNAL TO THE SYSTEM "A" OR "B" ANHONIA CONTROLLER FAULT LOGIC CIRCUIT FOR AUTOMATIC SWITCHOVER TO SECONDARY CONTROLLER AND CONTROL VALVE FOR UNDER TEMPERATURE PROTECTION. THE AMMONIA BOILER SYSTEM IS USED DURING POSTLANDING GPERATIONS, LAUNCH ABORTS, AND AS A BACKUP SYSTEM DURING NORMAL DEDRBITS.

PRINT DATE: 06/14/90 PAGE: 2 FAILURE MODES EFFECTS ANALYSIS (FMEA) -- CRITICAL FAILURE MODE NUMBER: 06-38-0403-01 2 D6/13/90 R REVISION SUBSYSTEM: ATCS - AMMONIA BOILER SYSTEM CRITICALITY OF THIS LRU : AMMONIA BOILER SUB-SYSTEM FAILURE MODE: IR3 ITEM NAME: TEMPERATURE SENSOR, ABS FAILURE MODE: OPEN (ELECTRICAL), OUT OF TOLERANCE/HIGH MISSION PHASE: DE-ORBIT DO VEHICLE/PAYLOAD/KIT EFFECTIVITY: 102 COLUMBIA DISCOVERY : 103 ATLANTIS : 104 ENDEAVOUR = 105 ■ CAUSE: VIBRATION, MECHANICAL SHOCK, CORRUSION ■ CRITICALITY 1/1 DURING INTACT ABORT ONLY? YES ABORT ONCE AROUND AOA RETURN TO LAUNCH SITE RTLS TRANS ATLANTIC ABORT TAL REDUNDANCY SCREEN A) PASS B) FAIL C) PASS PASS/FAIL RATIONALE: A) **= B**) RÉDUNDANCY SCREEN 'B' FAILS BECAUSE FAILED SENSOR CANNOT BE MONITORED BY CREW OR GROUND. - FAILURE EFFECTS -(A) SUBSYSTEM:

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LOSS OF UNDER TEMPERATURE PROTECTION FOR THE AFFECTED SYSTEM.

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■ (B) INTERFACING SUBSYSTEM(S): NO EFFECT UNLESS PRIMARY CONTROLLER LOGIC FAILS.

(C) MISSION: NO EFFECT.

PAGE:

- (D) CREW, VEHICLE, AND ELEMENT(S): NO EFFECT.
 - (E) FUNCTIONAL CRITICALITY EFFECTS:
 FUNCTIONAL CRITICALITY EFFECT LOSS OF ALL FREON UNDER TEMPERATURE
 CONTROL (PRIMARY CONTROLLER AND REDUNDANT AMMONIA SYSTEM PRIMARY AND
 SECODARY CONTROLLERS) CAN FREEZE THE INTERCHANGER AND RESULT IN RUPTURE
 OF WATER AND FREON COOLANT LOOPS. LOSS OF VEHICLE COOLING CAN CAUSE
 LOSS OF CREW/VEHICLE.

- DISPOSITION RATIONALE -

(A) DESIGN:
SENSOR IS A PLATINUM ELEMENT RESISTANCE THERMOMETER WHICH FUNCTIONS
AS PART OF A CLOSED LOOP CONTROL SYSTEM TO PROVIDE UNDER TEMPERATURE
PROTECTION TO KEEP THE INTERCHANGER FROM FREEZING. EACH AMMONIA
SYSTEM ("A" OR "B") USES A DEDICATED SENSOR. SENSORS HAVE SHEET METAL
COVERS TO PREVENT MOISTURE FROM ENTERING THE SENSOR ELEMENT. PLATINUM
SENSOR IS BONDED TO STAINLESS STEEL TUBING, WHICH IS COMPATIBLE WITH
FREON 21. OUTSIDE ENVIRONMENT. AND AMMONIA.

■ (B) TEST:

QUALIFICATION TEST — QUALIFICATION TESTED FOR 100 MISSION LIFE.

PERFORMANCE CHECK. VIBRATION TESTED AT 0.01 G**2/HZ FOR 48 MIN/AXIS

AND SHOCK TESTED AT +/- 20 G IN EACH AXIS.

ACCEPTANCE TEST - COMPONENT VENDOR ATP CHECKS CALIBRATION, INSULATION RESISTANCE. AMEMONIA BOILER SUBSYSTEM ATP CHECKS FOR LOW TEMPERATURE TRANSFER FUNCTION AND TRANSFER TEMPERATURES FOR SYSTEMS "A" AND "B".

OMRSD - AMMONIA CONTROLLER SWITCHOVER CHECKOUT EVERY TEN FLIGHTS.

(C) INSPECTION:
RECEIVING INSPECTION
RAW MATERIAL AND PROCESS CERTIFICATIONS VERIFIED BY INSPECTION. PARTS
PROTECTION VERIFIED BY INSPECTION.

CONTAMINATION CONTROL PROCESSES AND CORROSION PROTECTION PROVISIONS ARE

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VERIFIED BY INSPECTION.

ASSEMBLY/INSTALLATION
MANUFACTURING, INSTALLATION AND ASSEMBLY OPERATIONS, INCLUDING PARTS
PROTECTION, ARE VERIFIED BY INSPECTION.

CRITICAL PROCESSES
SPOT WELDING OF WIRES TOGETHER AND BRAZING ARE VERIFIED BY INSPECTION.

TESTING
ATP, INCLUDING INSULATION RESISTANCE AND LEAD WIRE PULL TESTS. IS VERIFIED BY INSPECTION.

HANDLING/PACKAGING

HA ING. PACKAGING AND STORAGE REQUIREMENTS ARE VERIFIED BY IT STION.

(' :ILURE HISTORY: N: 'LICABLE FAILURE HISTORY.

(E) JPERATIONAL USE:
FIRST FAILURE IS NOT DETECTABLE. WITH A SUBSEQUENT FAILURE (PRIMARY CONTROL LOGIC), CREW ACTION IS REQUIRED TO SWITCH TO THE SECONDARY CONTROLLER OR TO THE OTHER AMMONIA SYSTEM.

- APPROVALS -

RELIABILITY ENGINEERING: D. R. RISING DU: SCACO 7/2-103
DESIGN ENGINEERING: J. MORGAN

QUAL'TY ENGINEERING: M. SAVALA

NAS ELIABILITY

NAS UBSYSTEM MANAGER:

N JUALITY ASSURANCE:

D TO TO THE CONTROL OF THE CONTROL O